

Manoeuvre
Threshold

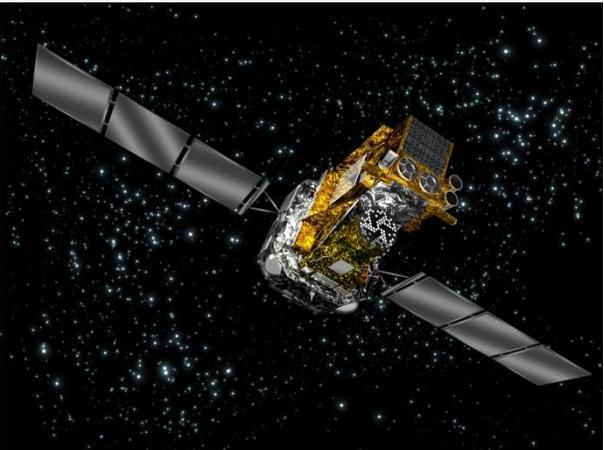
&
Strategy

Use of MASTER and ARES for Operational Collision Avoidance Support Planning

Master Modelling Workshop 2021
March 3, 2021

Francesca Letizia | Sven Flegel | Klaus Merz
ESA Space Debris Office

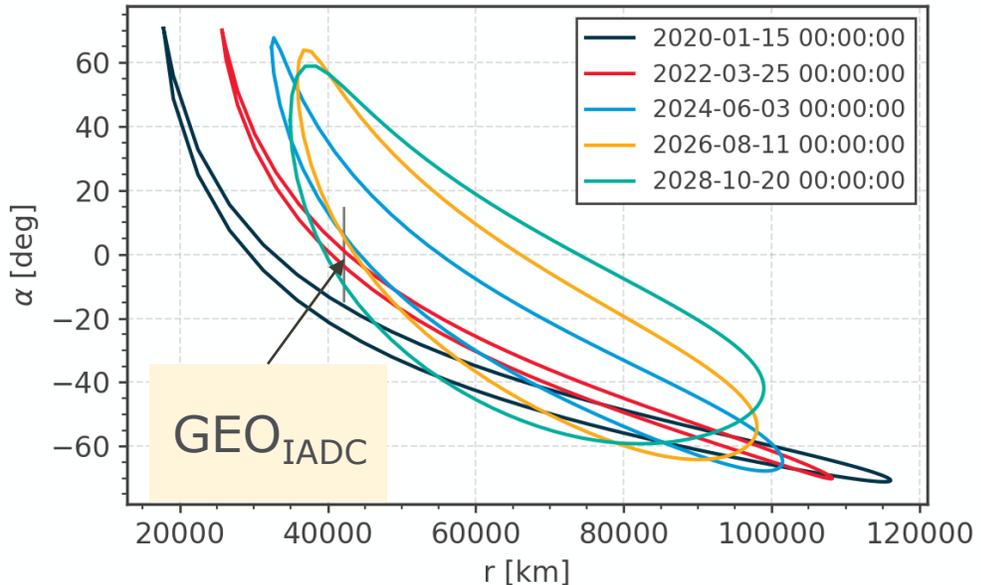
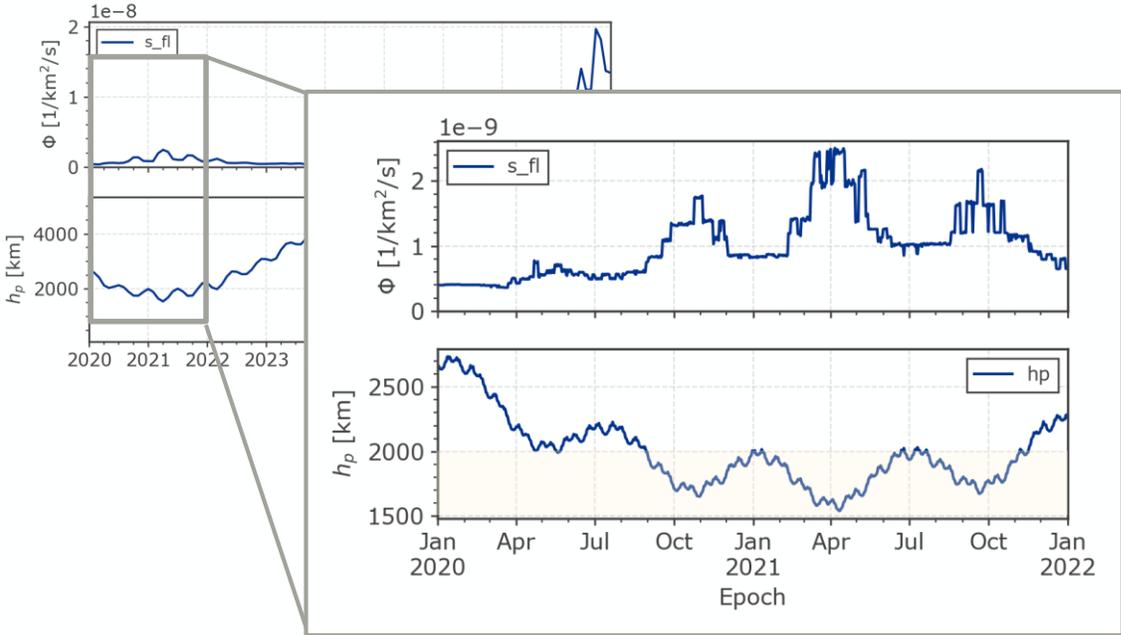
Collision Avoidance Analysis for HEO missions



Integral
LEO crossing
Sept 2020-Nov 2021



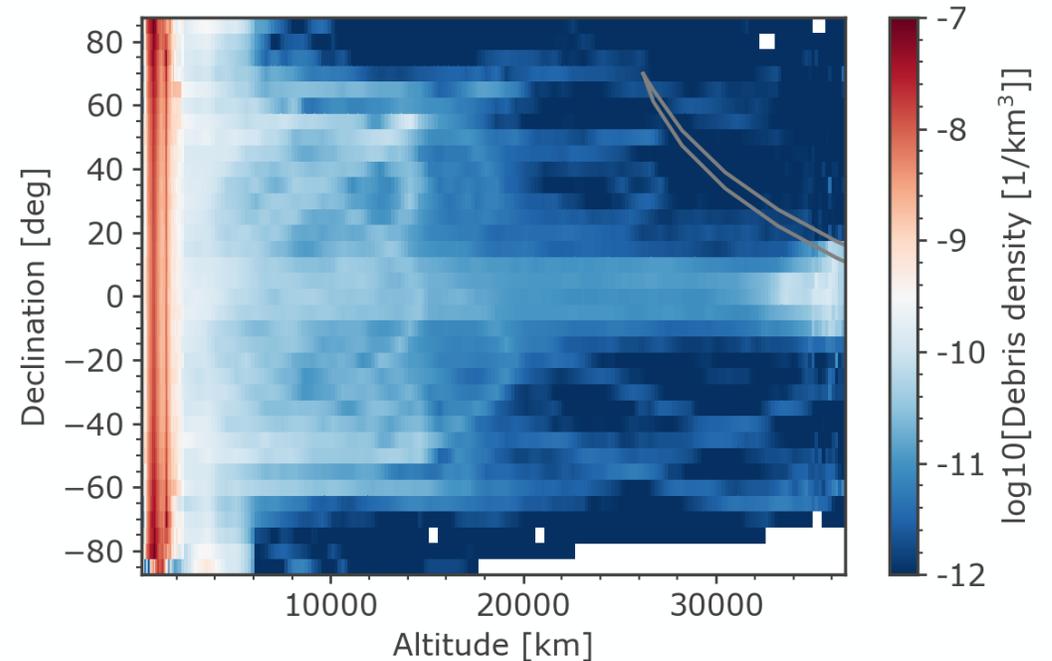
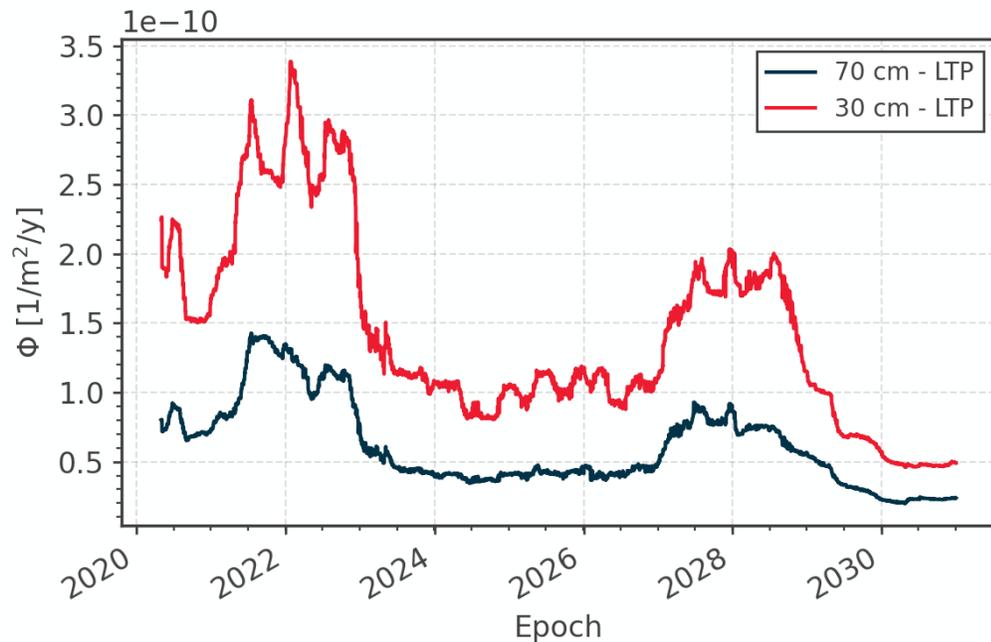
XMM
GEO crossing
mid 2020-mid 2029



Flux analysis – Example: XMM

Python wrapper for **analysis** along given **trajectory** files
Interface to common **orbit file types** (e.g. OEM, NAPEOS files)
Output: plots + .csv file

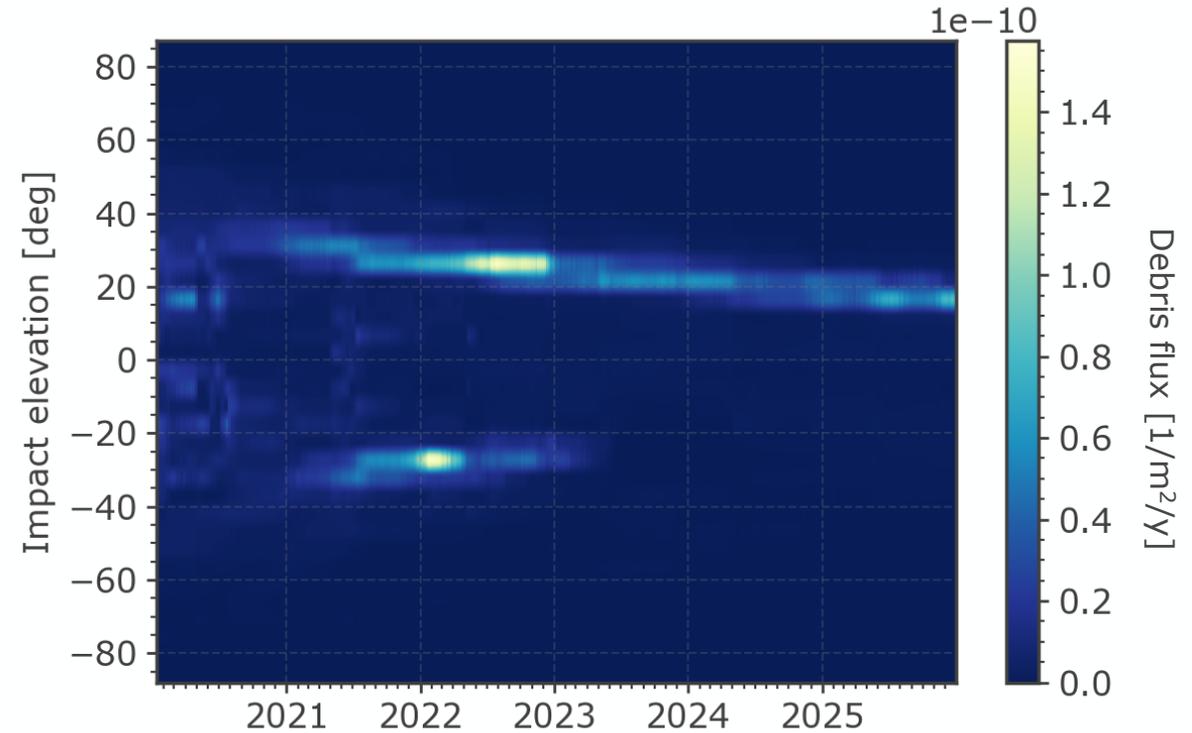
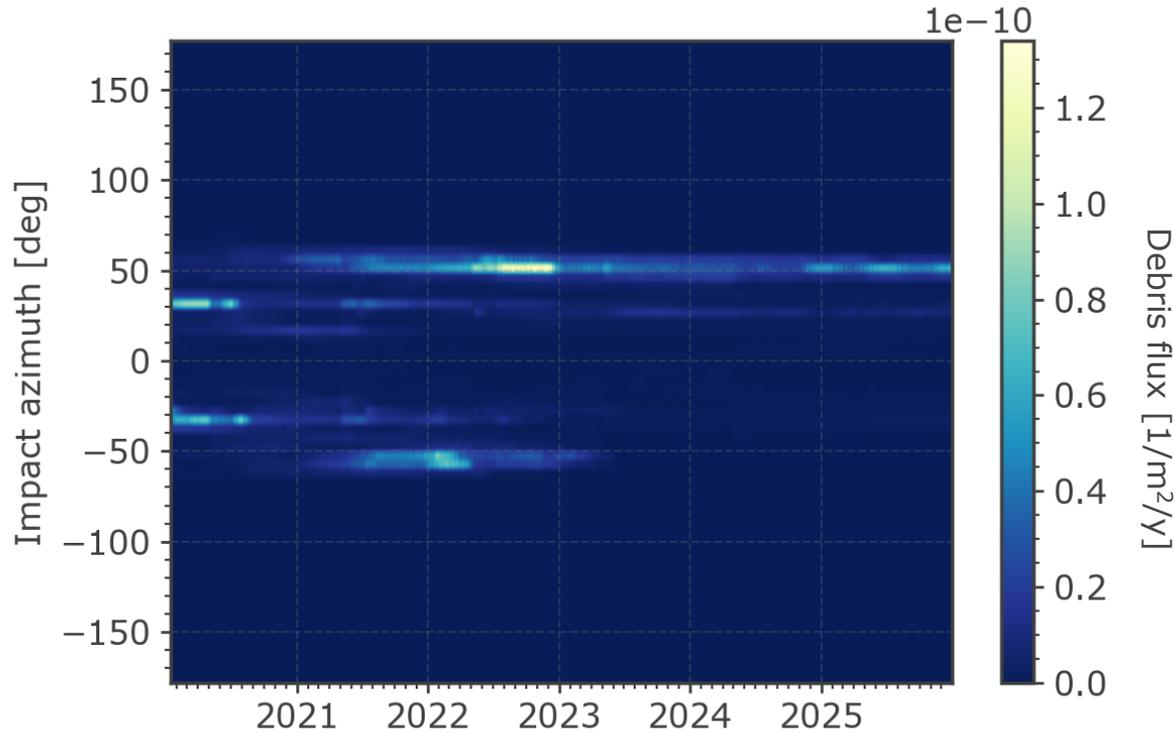
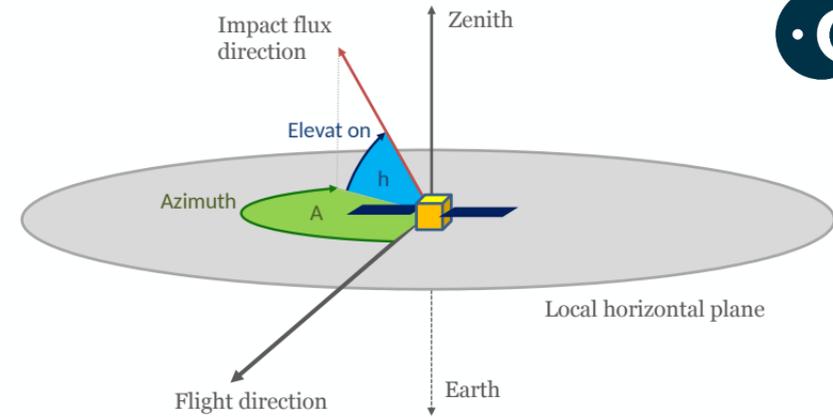
Debris flux



30-70 cm cut-off for detectability in GEO
LTP – Long-Term Propagation

Flux analysis – Example: XMM

Preliminary encounter **geometry** characterisation
in **azimuth** and **elevation**

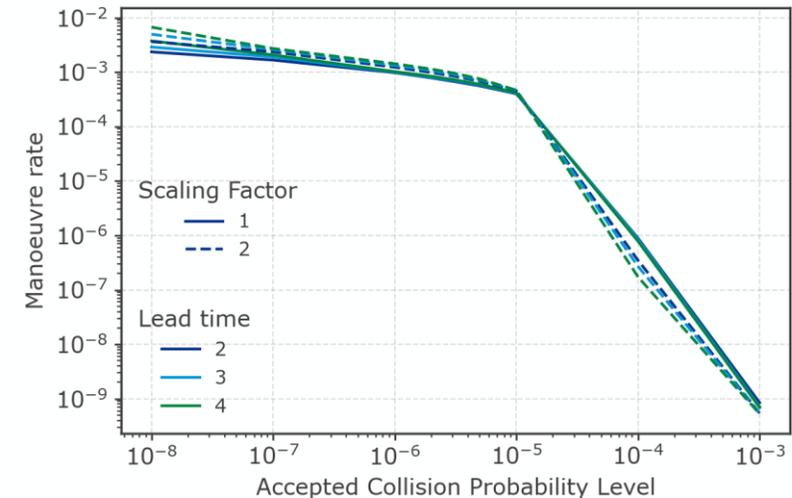
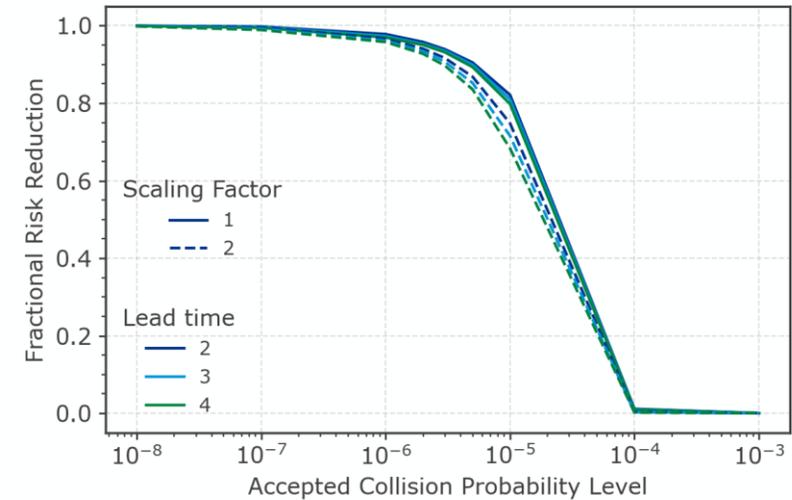


Collision probability & threshold definition

Next step: **DRAMA/ARES** analysis

- Check on the **mitigated risk** through collision avoidance at the epoch of **maximum flux**
- **Parametric** evaluation based on
 - Minimum debris size
 - Target covariance
 - Chaser covariance (i.e. scaling factor)
 - Lead time
- Definition of a suitable **reaction threshold** and determination of the corresponding **manoeuvre rate**

XMM





If manoeuvre (CAM) necessary:

- How much Δv ?
- What kind of CAM? e.g. Hohmann
- CAM schedule (e.g. time to TCA)



Require:

Target & chaser
states + uncertainties
for likely close encounters

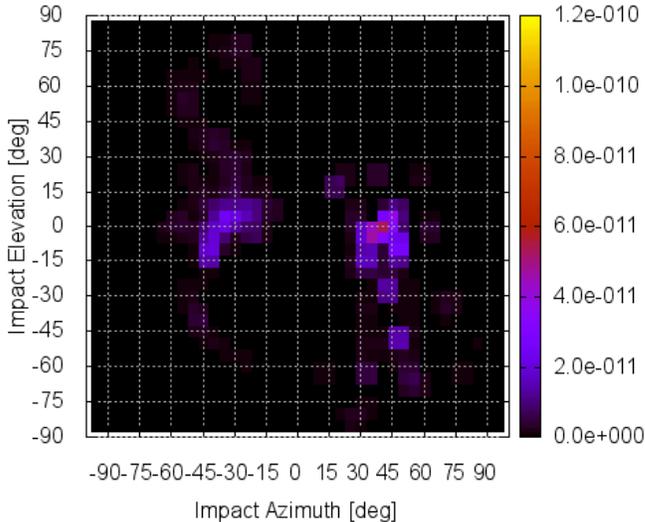
- Target orbit prediction (ESA flight dynamics team)
- Chaser size threshold



heat maps

INTEGRAL

ESA-MASTER Model v8.0.2
3D flux distribution vs. Impact Elevation and Impact Azimuth



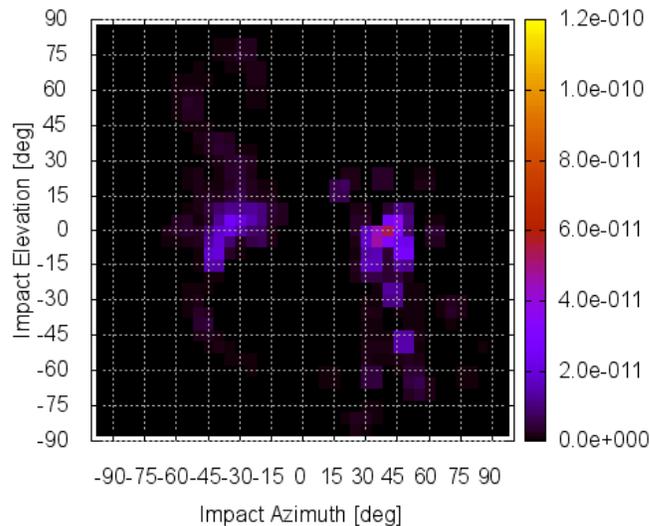
- Target orbit prediction (ESA flight dynamics team)
- Chaser size threshold



heat maps

INTEGRAL

ESA-MASTER Model v8.0.2
3D flux distribution vs. Impact Elevation and Impact Azimuth



Insight from 2-d heat maps

- Likeliest impact properties

Caveat

- Need to combine multiple heat maps for full picture
- Likely target & chaser state not readily available
- Likeliest per chaser category (e.g. orbit, size) not directly accessible

Realistic Close Encounters

- Target orbit prediction (ESA flight dynamics team)
- Chaser size threshold



cpe files



- GEO chasers
- MEO chasers
- LEO chasers

INTEGRAL



Realistic Close Encounters

- Target orbit prediction (ESA flight dynamics team)
- Chaser size threshold



cpe files

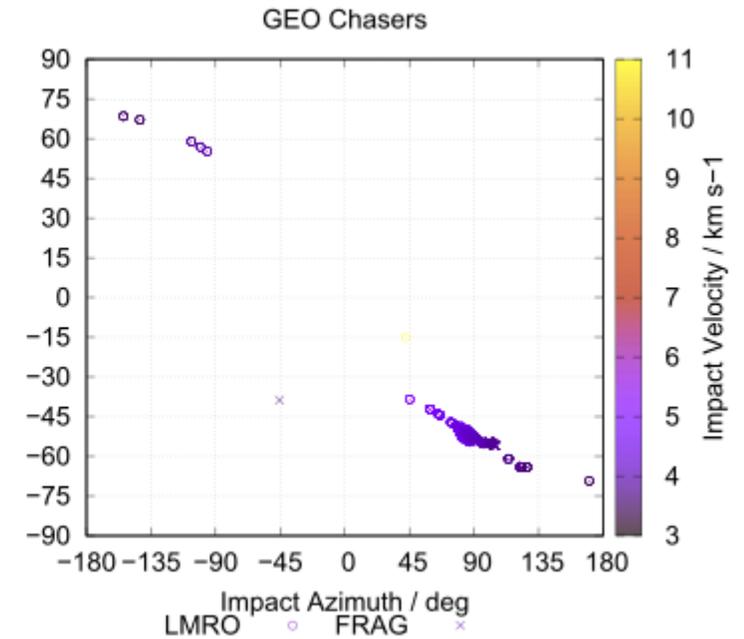
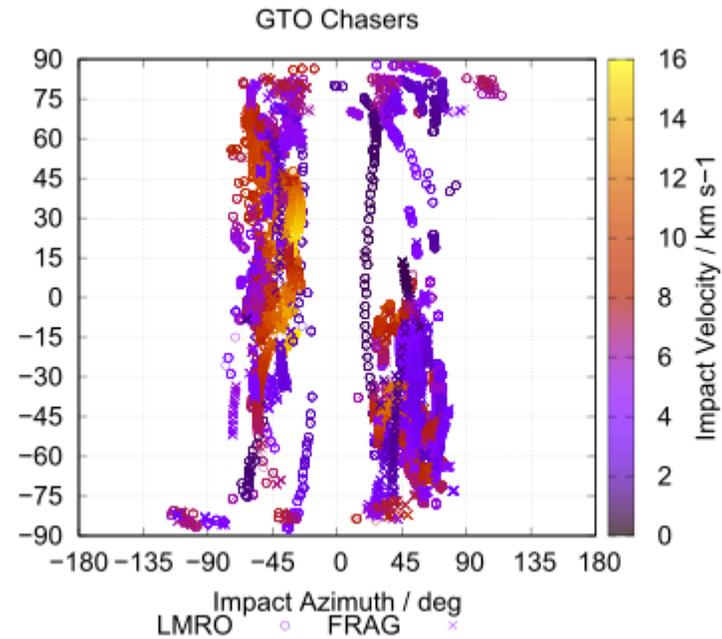
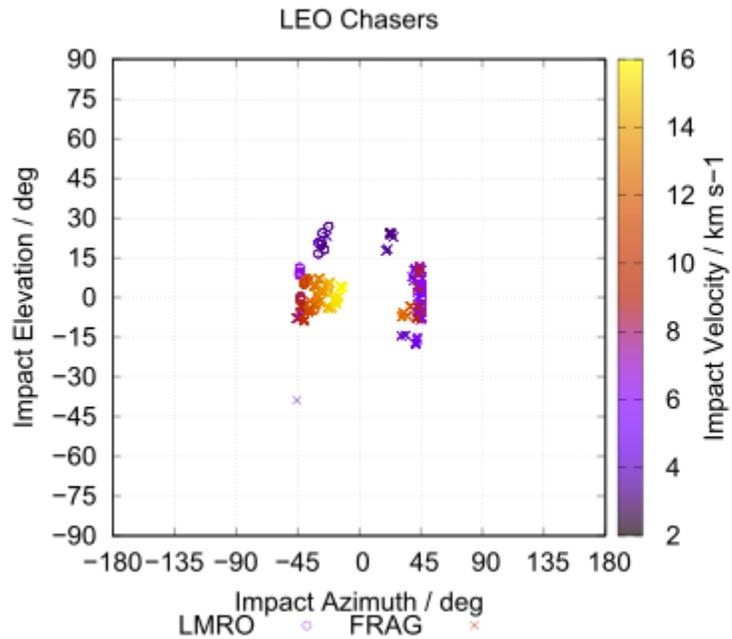


GEO chasers

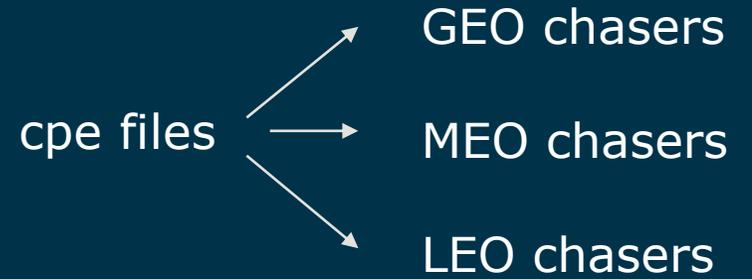
MEO chasers

LEO chasers

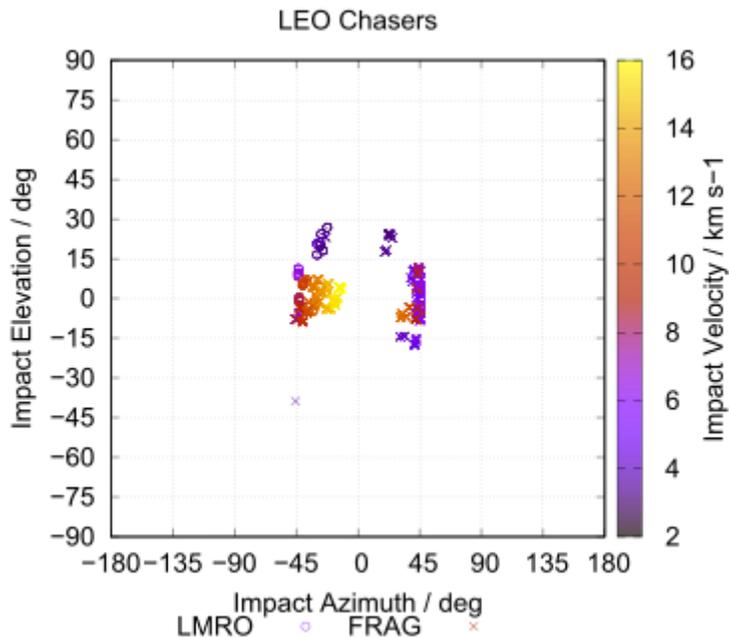
INTEGRAL



- Target orbit prediction (ESA flight dynamics team)
- Chaser size threshold



INTEGRAL



Insight

- Chaser orbit (sma, inc, ecc)
- Impact angle + Δv combination
- Impact location (important for ecc > 0)

Postprocessing

→ realistic & significant target & chaser states

Outlook / Summary

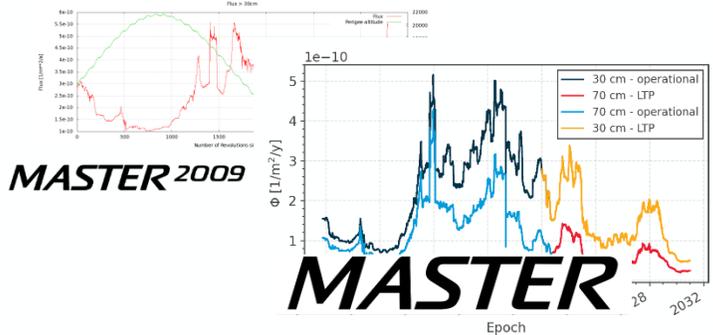


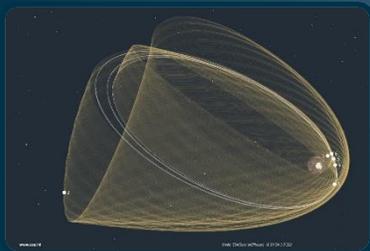
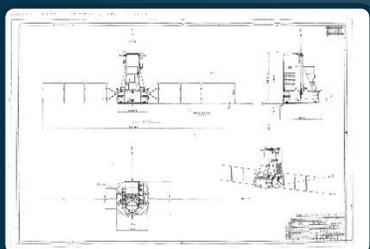
- Manual extract likeliest:
 - .. chaser per orbit (LEO/HEO/GEO)
 - .. impact location

- Interfaces:
 - **Python** wrapper
 - Options to load a **pre-computed trajectory** (with standard orbit format)

- Re-assessment for long-running missions can span over multiple model versions: More visibility of the **changes** in the **reference populations**

Rank	Target	Chaser





**Manoeuvre
Threshold**

**&
Strategy**

Use of MASTER and ARES for Operational Collision Avoidance Support Planning

Dr. Francesca Letizia | Dr. Sven K. Flegel | Dr. Klaus Merz
ESA/ESOC Space Debris Office (OPS-SD)
Robert-Bosch-Str. 5, 64293 Darmstadt, Germany
francesca.letizia@esa.int | sven.flegel@esa.int | klaus.merz@esa.int
<http://www.esa.int/debris>