

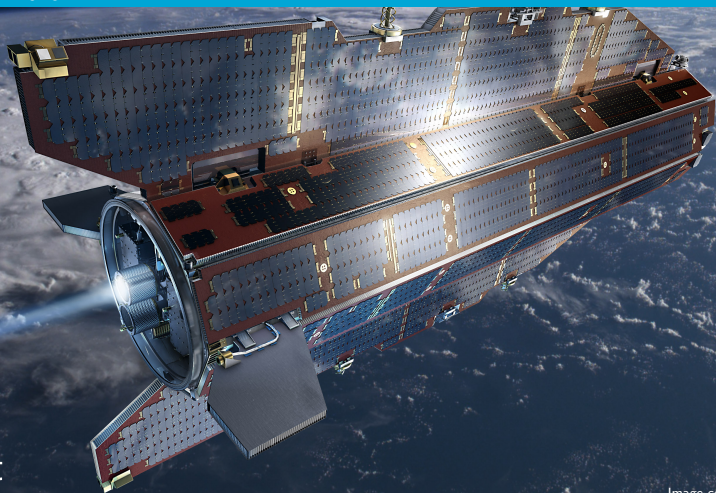
Aerodynamic oscillations during GOCE de-orbit

T. Visser (t.visser-1@tudelft.nl), J. Beck, E. Doornbos

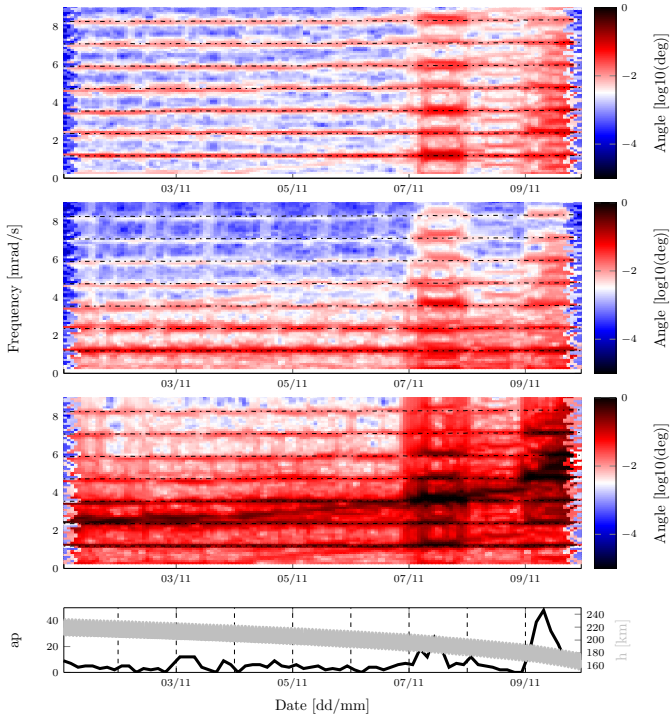
Delft University of Technology, Belstead Research Limited

4th International Space Debris Re-entry Workshop, ESOC, Darmstadt, Germany

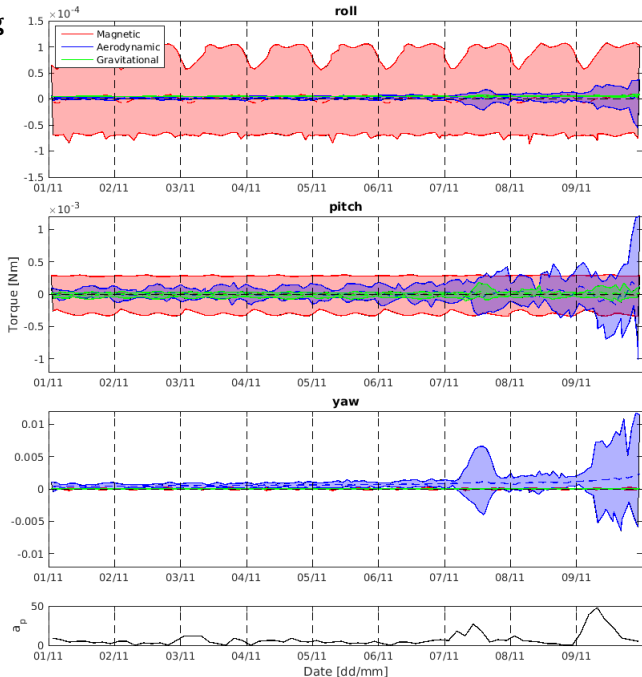
1 March 2018



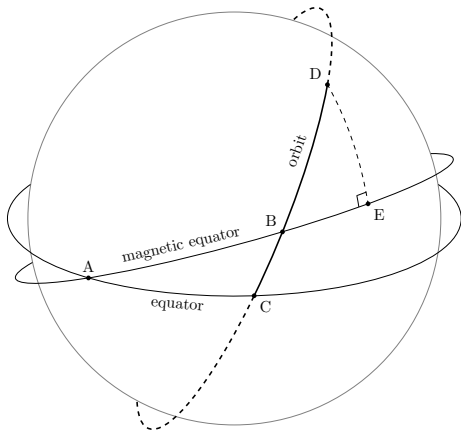
Euler angles: Will it tumble?



Torques during de-orbit



Simplified Earth model



Geomagnetic field:

$$B_N = B_0 \left(\frac{R_E}{R} \right)^3 \cos \delta_M$$

$$B_E = 0$$

$$B_D = 2B_0 \left(\frac{R_E}{R} \right)^3 \sin \delta_M$$

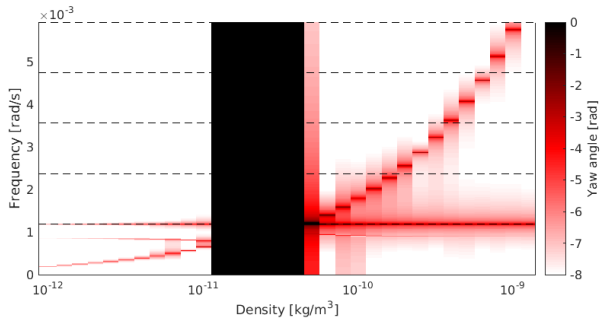
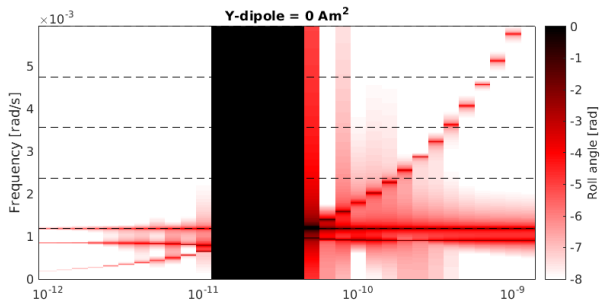
Atmospheric co-rotation:

$$\beta = -\frac{\dot{\tau} R}{V} \sin i \cos(\chi_0 + nt)$$

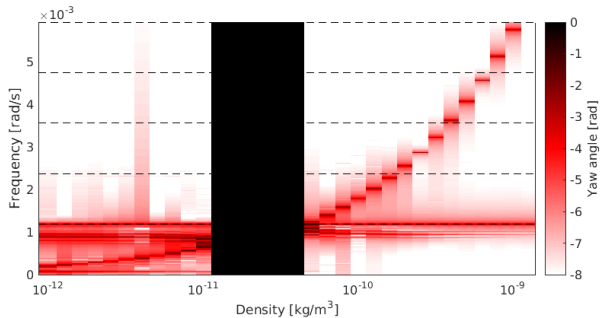
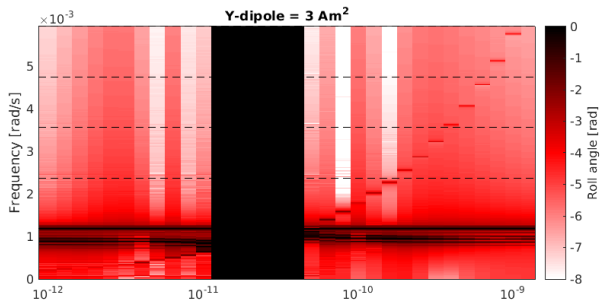
Linearized attitude model

	$\ddot{\varphi}$	$\ddot{\theta}$	$\ddot{\psi}$
φ	n T_M T_G	T_A T_M	T_M
θ	T_A T_M	T_A T_M T_G	T_M
ψ	T_M	T_M	T_A T_M
$\dot{\varphi}$			n
$\dot{\theta}$			
$\dot{\psi}$	n		
forcing	T_M	T_M	T_A T_M

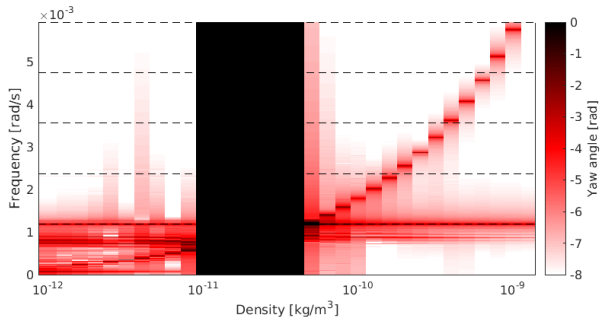
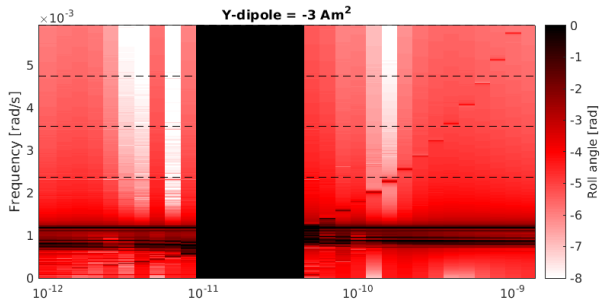
Density



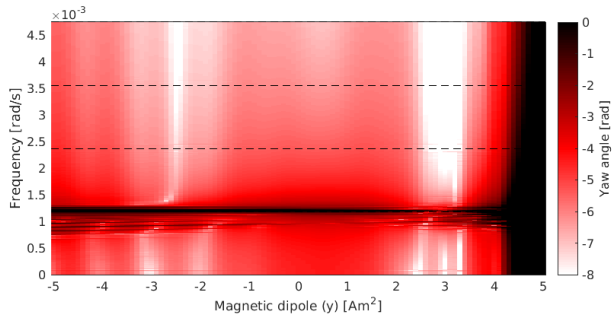
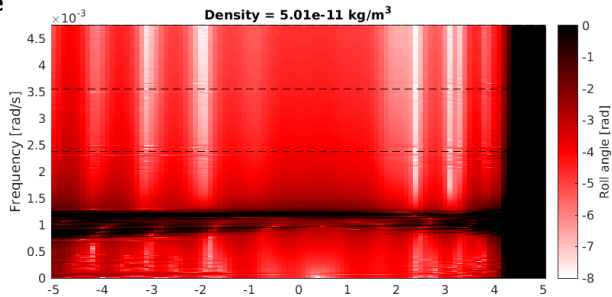
Density



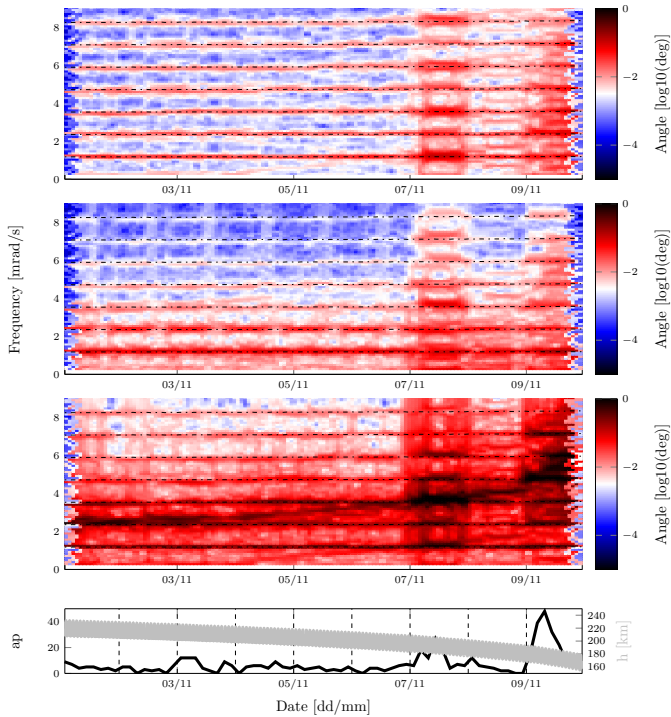
Density



Magnetic dipole

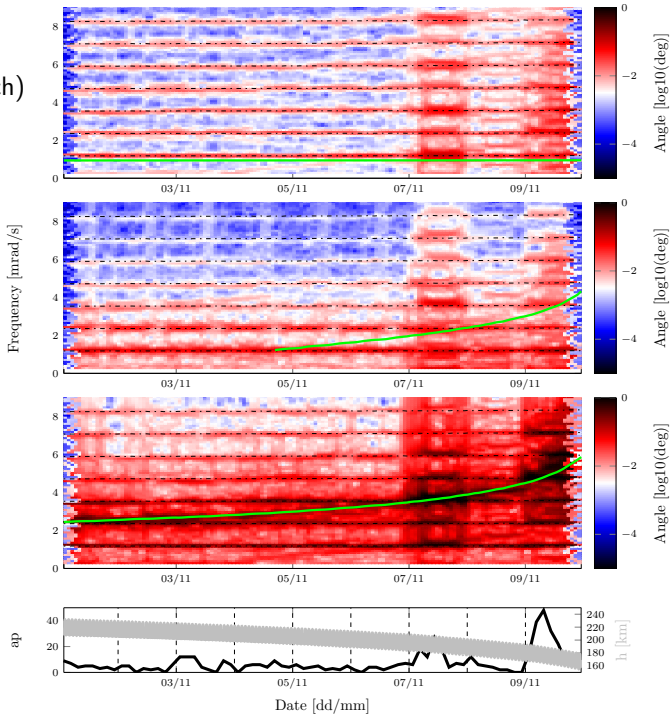


Back to reality



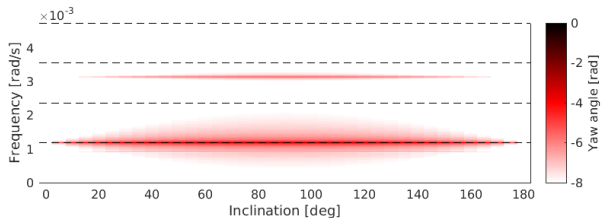
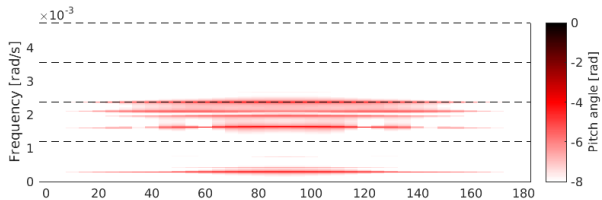
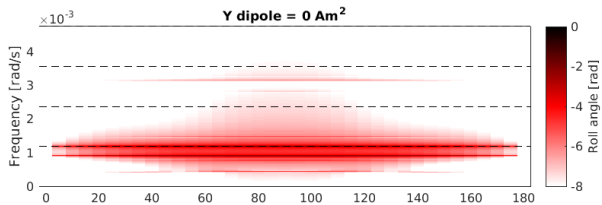
Back to reality

- NRLMSISE-00
- SPARTA (G.March)



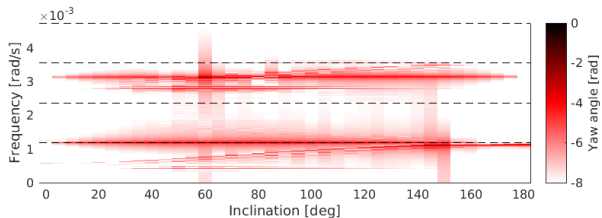
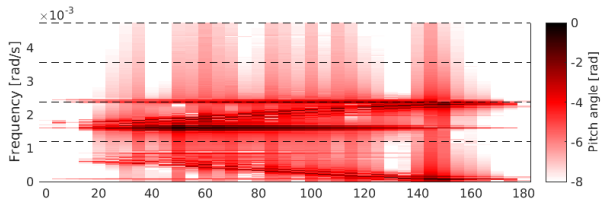
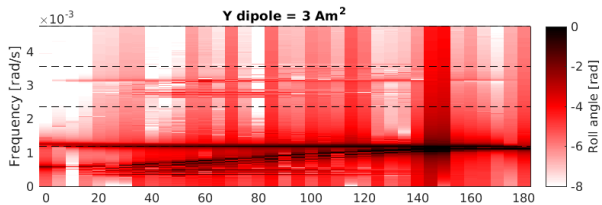
Inclination

$$3 \times 10^{-10} \text{ kg/m}^3$$



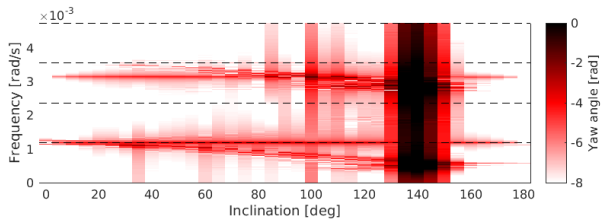
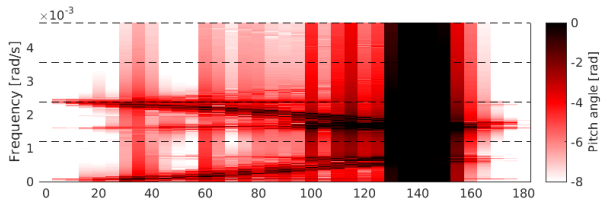
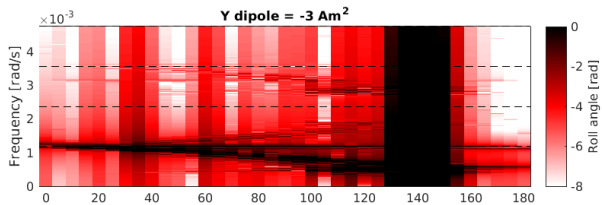
Inclination

$$3 \times 10^{-10} \text{ kg/m}^3$$



Inclination

$$3 \times 10^{-10} \text{ kg/m}^3$$



Conclusions

- ▶ Significant magnetic torques during de-orbit;
- ▶ High density required for pitch stability;
- ▶ Resonance destabilizes attitude.

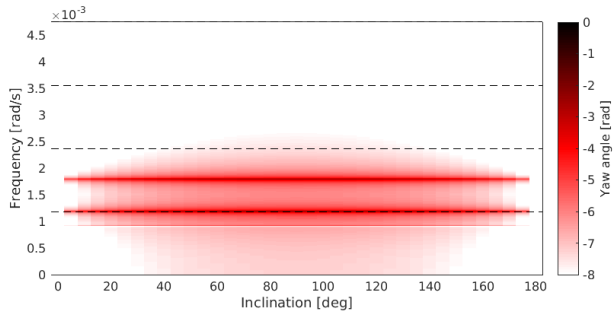
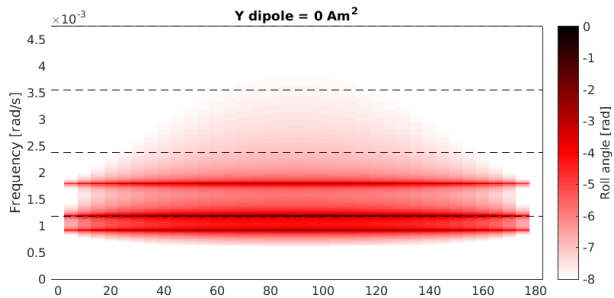
Recommendations

- ▶ Include $C_{l_{\alpha,\beta}}$ (roll aerodynamics);
- ▶ Analytic expressions for aerodynamic coefficients (B. Fritsche);
- ▶ Verification using simulations.

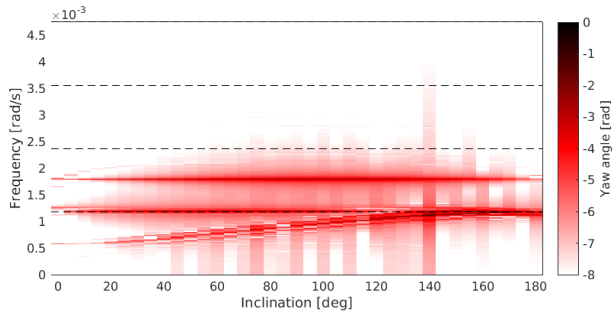
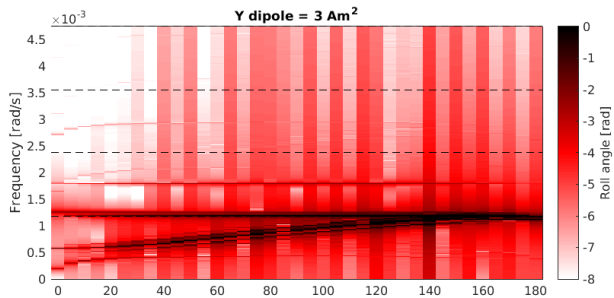
T. Visser: t.visser-1@tudelft.nl

EXTRA SLIDES

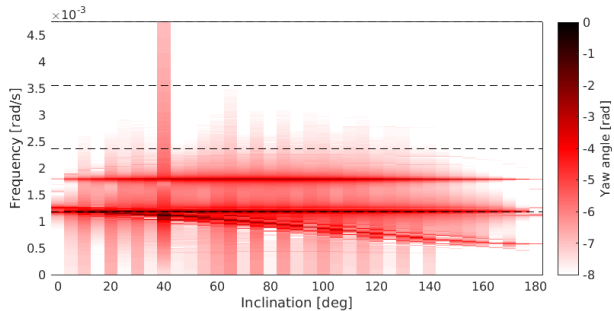
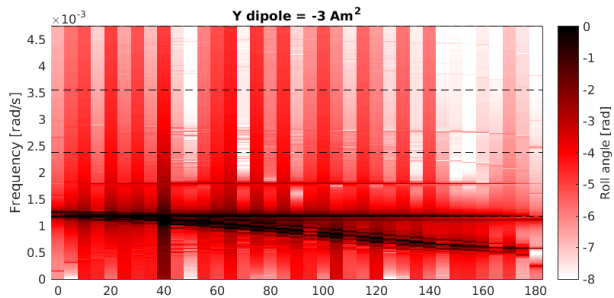
Inclination



Inclination

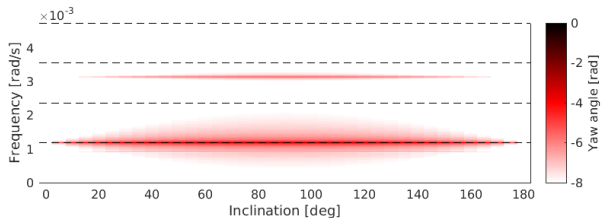
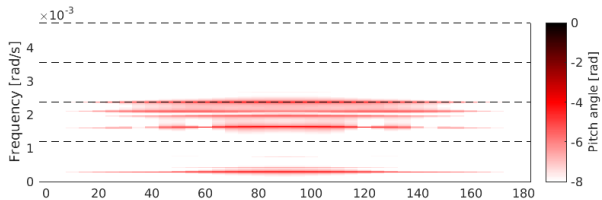
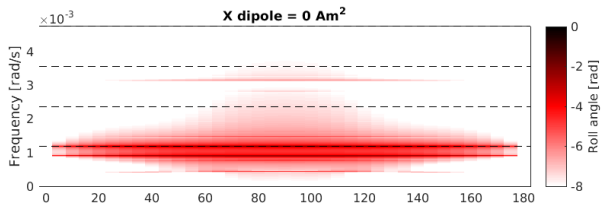


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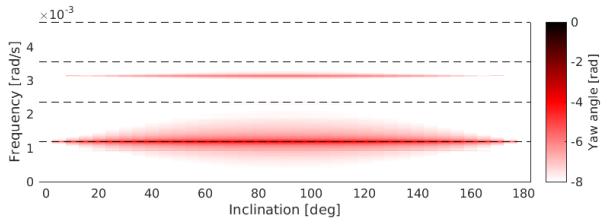
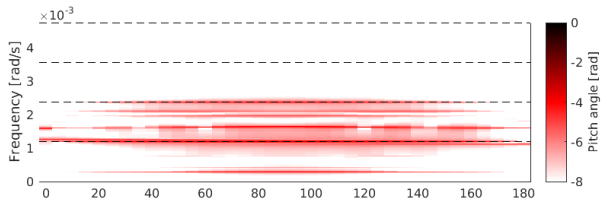
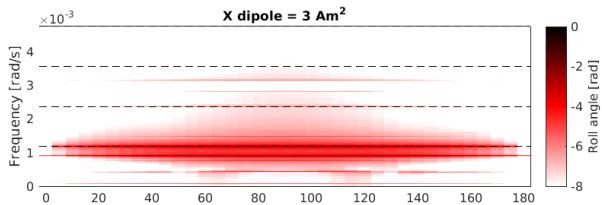
Inclination

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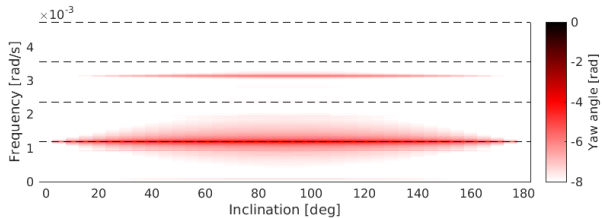
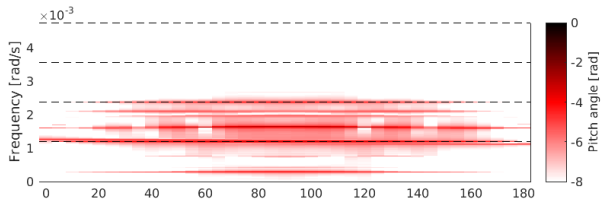
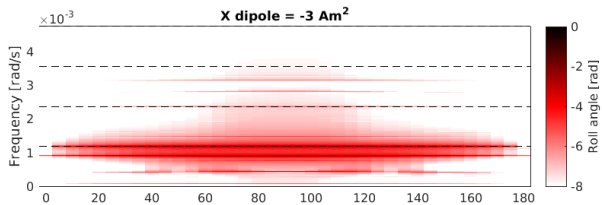
Inclination

$$3 \times 10^{-10} \text{ kg/m}^3$$



Inclination

$$3 \times 10^{-10} \text{ kg/m}^3$$



Control dipole

$$\mu_x = -\mu_c \cos(2nt)$$

